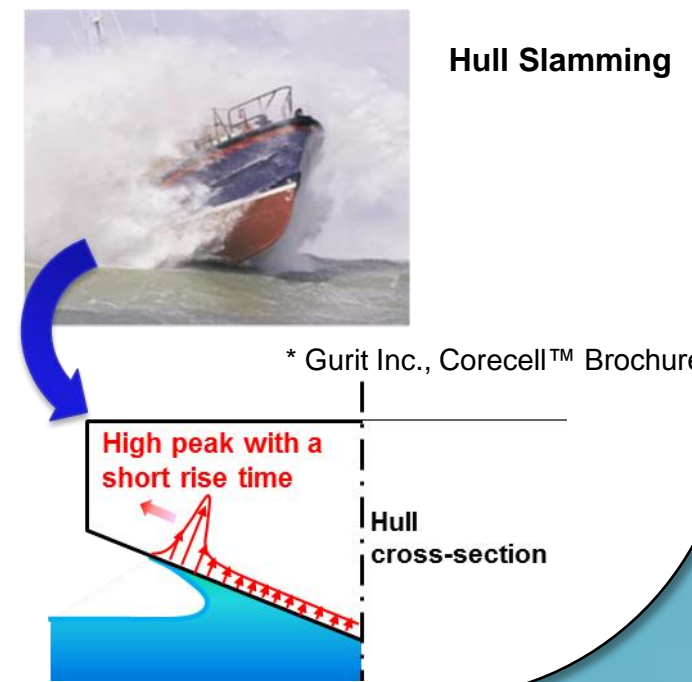


Y. Tsuchida (PhDCE), and J.W. Gillespie, Jr.

University of Delaware . Center for Composite Materials . Department of Civil and Environmental Engineering

INTRODUCTION

- ◆ Advanced composite sandwich structure is widely used as the primary hull structure of high-speed marine vessels.
 - ◇ Structural sandwich composites with foam cores have proven advantages.
 - Improved stiffness / strength
 - Lightest structural material form
 - Highest specific properties
- ◆ Future trend in marine industry requires that the reliability and durability have to be enhanced so that composite sandwich hulls are utilized more safely and effectively.
- ◆ Slamming is a crucial concern in the structural design of a high-speed marine craft.
 - ◇ High pressure peak causes local material or structural degradation.
 - ◇ High frequencies of slamming events increase fatigue failure.



RESEARCH QUESTIONS AND PROPOSED APPROACHES

- ◆ Material Behavior of Composite Sandwiches due to Slamming
 - ◇ Focus on the characterization of the dominant foam core by lab-scale testing considering multiple slam events.
 - ◇ Proposed method approach is to utilize **Dynamic Mechanical Analysis (DMA)**.
- ◆ Prediction of Slamming Loads and Structural Response
 - ◇ Focus on the CFD and FEA modeling of a composite sandwich structure incorporating the foam core property subjected to wave slamming-loads.
- ◆ Marine Design
 - ◇ Estimate the long-term effect and degradation of composite sandwich hull material due to slamming.

BASELINE SANDWICH CONSTRUCTION

◆ Carbon/Epoxy Face Sheets

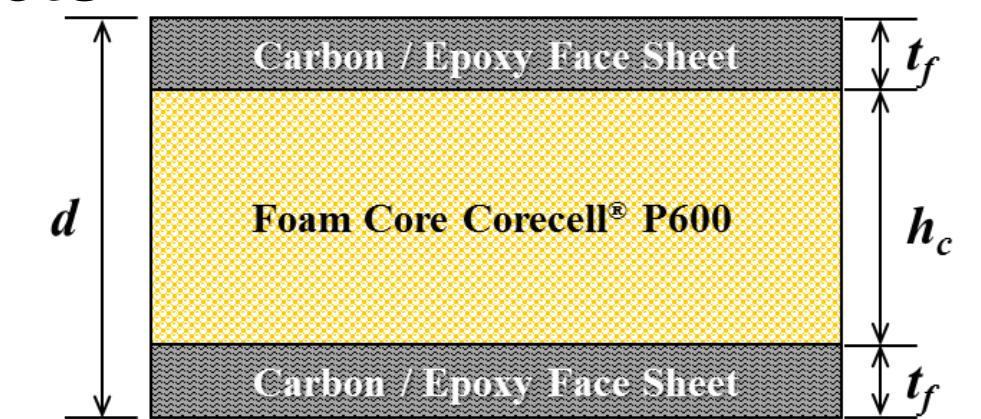
- Eight (8) layers of 4x4 Twill Carbon Fabric
- $t_f = 2.4 \text{ mm}$ [0.1"]
- $\rho_f = 1478 \text{ kg/m}^3$

◆ Foam Core

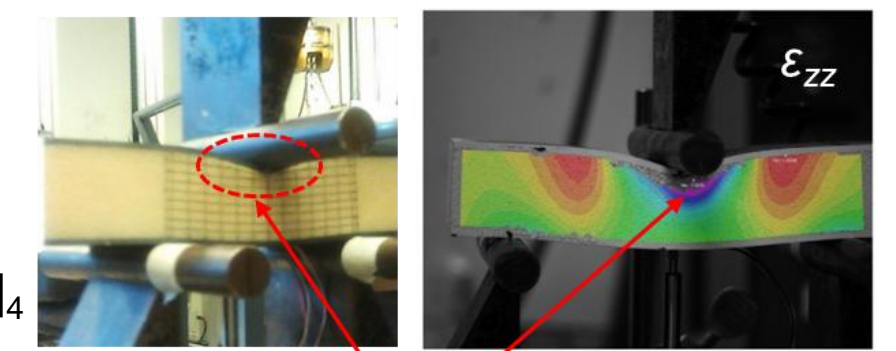
- Styrene Acrylonitrile (SAN) Polymer
- $h_c = 38 \text{ mm}$ [1.5"]
- $\rho_c = 122 \text{ kg/m}^3$

◆ Processing

- ◇ Out of Autoclave (OOA): Prepreg Vacuum Bagging / Oven Cure
- Quasi-isotropic Layup: $[(0/90)/(+45/-45)]_4/\text{core}/[(-45/+45)/(90/0)]_4$
- No Adhesive Interface
- Debulked at RT for 12 hrs
- Cured at 65°C [150°F] for 16 hrs



◆ Typical Failure under Three Point Bending



BENCHMARK TESTING OF FOAM CORE

Static Testing (ASTM D1621)

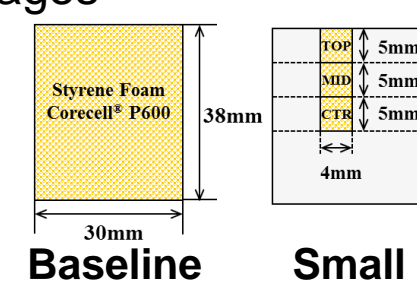


DMA



◆ Verification in Static Tests

- Mechanical properties and micro damages
- Specimen dimension sensitivity
- Homogeneity through the thickness



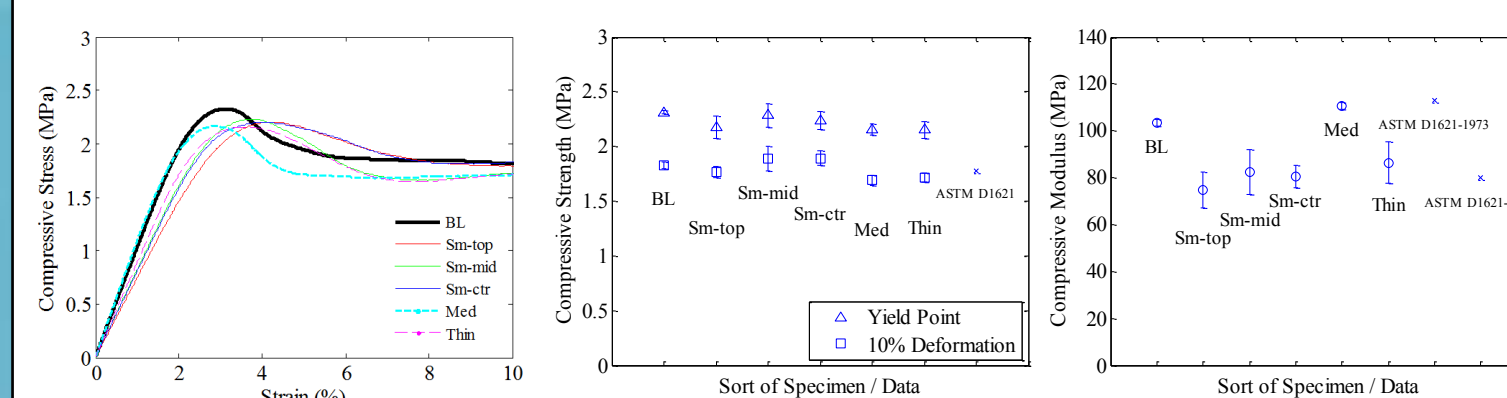
◆ Test Specimen

- Cuboid, square in cross section, different in dimensions.

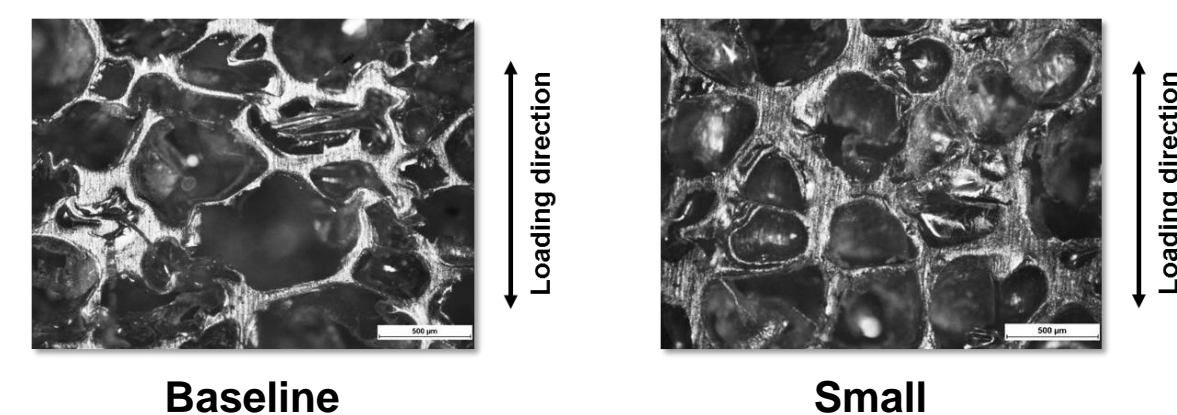
	Baseline	Small (DMA-type)	Medium	Thin
W, T	30 mm	4 mm	15 mm	30 mm
H	38 mm	5 mm	19 mm	5 mm
Slenderness Ratio	4.40	4.33	4.39	0.58
Strain Rate	0.1	0.1	0.1	0.1

RESULTS OF STATIC COMPRESSIVE TESTS

◆ Comparison of Static Mechanical Properties



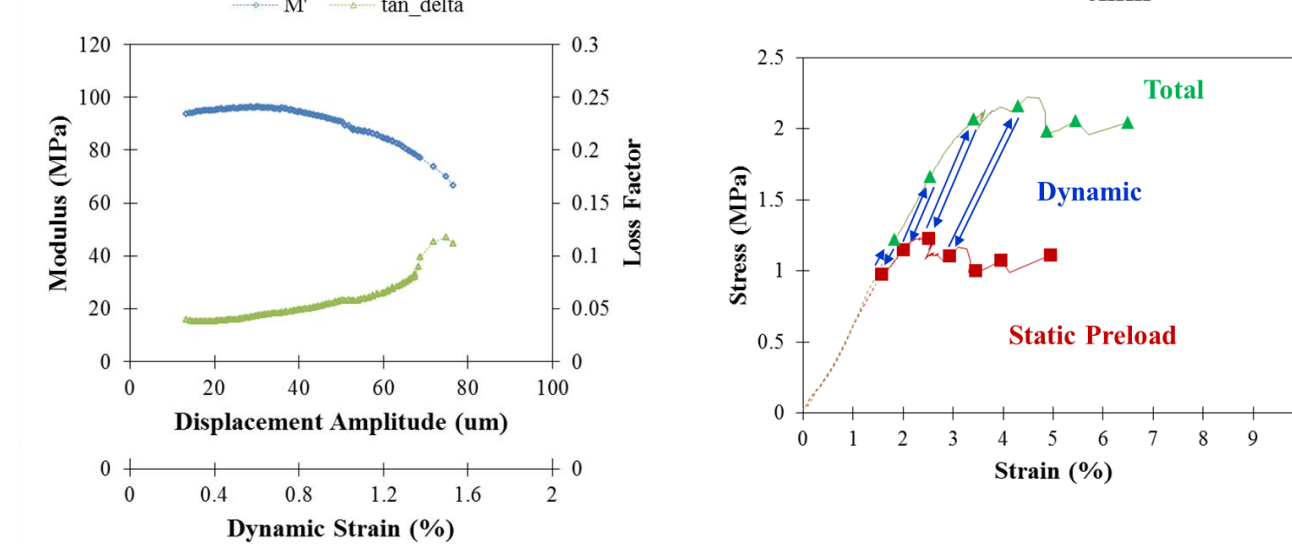
◆ Microscopic Failure



PRELIMINARY DMA EXPERIMENTS

◆ Dynamic Strain Sweep Testing

- ◇ Increase the displacement amplitude until a foam crushes.
- ◇ At ambient RT with 1 Hz
- ◇ Small (DMA-type) specimens



SUMMARY

- ◆ This research proposes utilizing DMA to investigate the fatigue property of a foam core under slamming events.
- ◆ The study uses the prospective sandwich construction with high impact tolerant styrene foam core and OOA prepreg processing.
- ◆ Static compression tests confirms small DMA-type specimens represent the consistent properties to the baseline foam material.

ACKNOWLEDGEMENTS

This work is supported by the Office of Naval Research through the Advanced Materials Intelligent Processing Center.