

FINITE ELEMENT ANALYSIS OF COMPOSITE PARTS

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OBJECTIVES

The objective of the finite element analysis of composite parts is to achieve a solid and sound product that will meet or exceed all the required performance criteria.

COMPOSITE PARTS



- Hummer Hood
- M35A3 hood
- M939 truck's Fenders
- Bulkhead
- Trailer



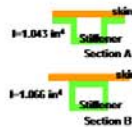
RE-ENGINEERING AND RE-DESIGNING OF M35A3 HOOD

Static Load – The hood shall withstand a 250 lb minimum static load applied over a 10 inch maximum by 10 inch maximum area on the center and on the front of the hood with an elastic deflection of not more than 0.5 inch at +250°F and not more than 0.25 inch at -50°F. The load shall be applied on the outside surface and the deflection measured at the center within 0.5 inch of the applied area.

Flexural Strength – The hood shall be mounted on a vehicle equivalent fixture. The load to lift the front right corner of the hood 0.375-0.500 inch, with center and right side of the hood secured, shall be 50 lb minimum. The load shall be applied at the center ±0.50 inch of the handle.

SHELL MODELED STIFFENER VS. BEAM MODELED STIFFENER

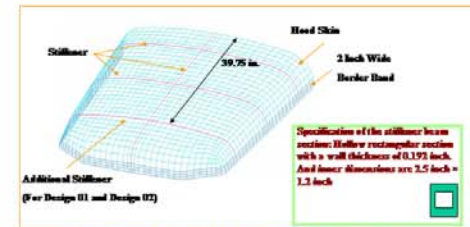
From simulation, we found there is no significant difference in the shell and beam element models, so beam elements could be used to model the hood stiffener. The reason behind using beam elements in the model is easy change of stiffener location and cross section.



	Shell Stiff	Beam Stiff
Upper/lower, Top-bottom	0.901	0.903
In Plane X stress (psi)	Min: 1198.0	988.0
	Max: 4822.8	455.0
In Plane Y stress (psi)	Min: 355.0	714.0
	Max: 455.0	488.0
Inter Laminar Shear Stress	25.87	N/A
In Plane Strain	0.836 %	0.836 %

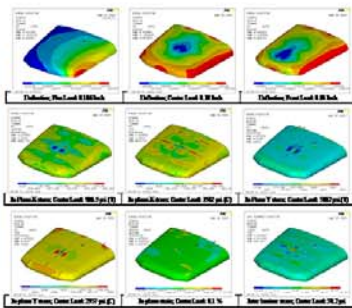


HOW WE CAN IMPROVE THE EXISTING DESIGN ?



Design	Hood skin: # Layers	Stiffener: # Layers	Total Weight:
Re-engineered:	Hood skin: 8 Layers	Stiffener: 8 Layers	62.55 lbs
Design 01:	Hood skin: 4 Layers	Stiffener: 8 Layers	38.84 lbs
Design 02:	Hood skin: 6 Layers	Stiffener: 4 Layers	45.55 lbs
Design 03:	Hood skin: 4 Layers	Stiffener: 4 Layers	33.89 lbs

RESULTS



RESULTS

MAX STRESS-STRAIN DATA SHEET: DESIGN 02

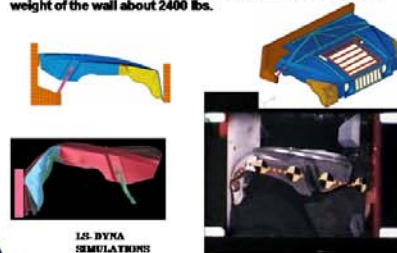
Material failure parameter	Standard value	Failure value/Design goal	Factor of safety
Deflection, in. (Flex Load)	0.106	0.375 - 0.5	
Deflection, in. (Center Load)	0.100	0.5	5.0
Deflection, in. (Front Load)	0.000	0.5	0.3
In plane X stress (psi)	Tension 906.5	50000.0*	64.0
	Compression 1682.0	56500.0*	36.7
In plane Y stress (psi)	Tension 3062.0	50000.0*	16.3
	Compression 2937.0	56500.0*	19.2
In-plane strain (%)	0.10	0.50	5.0
Minimum Inter Laminar Shear (Front/Back) stress (psi)**	30.2	4500.0**	149.0
Through the thickness stress	No data available from simulation		
Impact determination	Simulation haven't been done		
Weight	45.55 lbs	N/A	N/A

10334035: In-plane stress and strain, and Inter-Laminar stress are the maximum of the load case.
*Refer values under 2008, Section Composite materials design and applications, 2008-1-1074-0104, p. 44
CSM Available, *Free edge interlaminar stress recommended for accuracy

TABLE 06: MAXIMUM STRESS-STRAIN DATA SHEET FOR DESIGN 02

RESULTS OF 30 MPH CRASH BEHAVIOR HMMWV HOOD

- Back wall was moving forward with a velocity of 30 MPH.
- The density of the back wall was increased to make the total weight of the wall about 2400 lbs.



I.S. DYNA SIMULATIONS

EXPERIMENT

Buckling comparison between simulation and experiment.

CONCLUSIONS

- Successfully simulated the crash behavior of the HMMWV hood.
- Re-designed the existing M35A3 hood for light weight and better performance.
- Completed the study of the effect of layer numbers and effect of stiffener location on the M35A3 hood performance.
- FEA was applied successfully (waiting through testing) to assess the stress-strains in the M35A3 hood model.

ACKNOWLEDGEMENTS

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